

# **PCEC in ACE**

The Power of Third Parties

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## **Abstract**

In 2014, NASA released the Project Cost Estimating Capability (PCEC). This is the continuation of data collection, methodology development, and model development activities that were formerly contained within NAFCOM. PCEC contains an accessible library of NASA cost estimating relationships (CERs) to enable analysts the flexibility to use or modify equations as needed for their analysis.

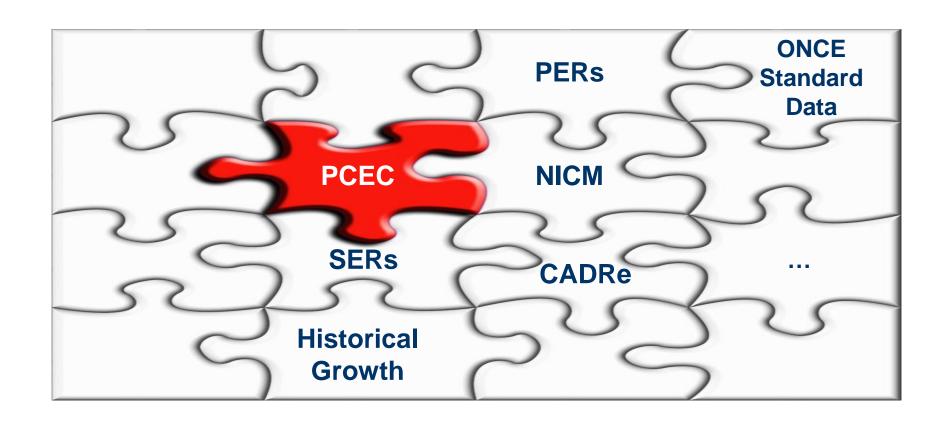
ACEIT is a Government funded application that is standard within the Army, DHS, Navy, and Air Force organizations. ACEIT consists of a suite of analyst applications ranging from a statistics application (CO\$TAT) to a time-phased data integration and simulation engine (ACE) to an integrated cost/schedule risk tool (JACS) that can be used for JCL analysis. The ACE application is robust and fully allows users to build their own models, incorporate phasing, and apply uncertainty to develop risk-based and time-phased cost estimates. In addition, ACE allows the user to assess/develop multiple scenarios on changes of input parameters or equations and to view the results in powerful Excel reports through the use of an Excel add-in (POST).

This paper demonstrates the ability for ACE to capture the equations contained in PCEC and develop estimates. In addition, the power of ACE allows for time-phasing of costs and for the incorporation of risk analysis based on simulation modeling. ACE's capability allows uncertainty on input parameters (e.g., mass) as well as to describe the error in the underlying cost equation. This paper will showcase the ease in which a PCEC based cost estimate can be run in ACE.



NASA's PCEC and ONCE frameworks empower analysts through enhanced visibility and transparency to build quality models and improves NASA's overall estimating and modeling capability





### ONCE is repository for quality information for analysts

### Analysts can use the information to build custom models

- PCEC equations
- NICM Results
- Analogies
- Schedule Estimating Relationships
- Phasing Relationships
- Mass Growth



## Robust API

## Learning Curve

## **Time Phasing**

Schedule Risk Analysis

Date calculations

Interactive Charts

**Excel**Integration



**Documentation** 

Automated Error Checking

Distribution specification

Unlimited What-If's

**Joint Confidence Level** 

Simulation

Tabular reports

### **Inputs**

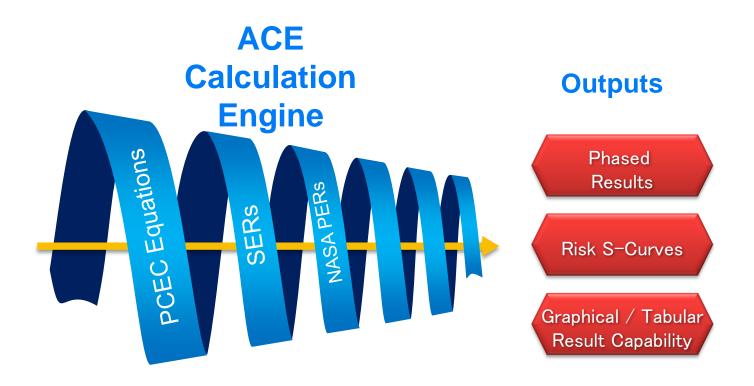
System Parameters

Subsystem Mass
& Technical

Schedule & Phasing

Instrument Costs

Risk/Uncertainty Inputs



ACE exists in a "stand-alone" state which can be used by analysis familiar with the platform.

The excel interface simply deliverers the parameters needed for ACE to complete the calculations and provide the data for Graphical and/or Tabular result generation.







# **Global Inputs**

#### PCEC INPUTS

Global Variables									
Multivariable Type	(1) Other								
System Test Hardware Quantity	1								
Flight Unit Percent	130								
Start Year	2013								
Flight Year	2020								
Design Life (# of months)	70								

Weight Tak	ole
Total Spacecraft	361.6
Structures & Mechanisms	226.1
Thermal Control	11.4
Reaction Control Subsystem	0
Electrical Power and Distribution	69.9
Command, Control & Data Handling	41.4
Attitude Determination & Control	12.8
Apogee Kick Motor	0

Wrap Values								
Fee Percent	0.1							
Program Support Percent	0.15							
Contingency Percent	0.05							
Vehicle Level Integration Percent	0.1							

System Integration								
	Hardware Class	Orbit	Spacecraft Class					
N	Crewed	Crewed	Crewed					
N	Crewed Earth Orbiting Observatory							
N	Engine	Engine	ne Engine					
N	LV	LV	Liquid Stage					
N	LV	LV Solid Stage						
Υ	Uncrewed	Earth Orbiting Communication						
Υ	Uncrewed	Earth Orbiting	Mapping/Meteorological					
Υ	Uncrewed	Earth Orbiting	Observatory					
Υ	Uncrewed	Earth Orbiting	Positioning					
Υ	Uncrewed	Earth Orbiting	Reconnaissance					
Υ	Uncrewed	Earth Orbiting	Scientific					
N	Uncrewed	Planetary	Inner Planet Explorer					
N	Uncrewed	Planetary	Lander					
N	Uncrewed	Planetary	Outer Planet Explorer					
N	Uncrewed	Planetary	Probe					

#### SCHEDULE ESTIMATE INPUTS

Select SER method:	QuickCost								
Please enter Phase A duration	40								
(months)	10								
QUICKCOST INPUTS									
Destination	Earth Orbiting								
Instrument Complexity	0.5								
Desired Confidence Level	70%								
·									
MCR INPU	MCR INPUTS								
Imaginary Intelligence (IMINT)	Ma								
Remote Sensor?	No								
Beginning of life (BOL) power in	624								

New or Replacement

All, Planetary, or Nonplanetary?

New

# **Subsystem Inputs**

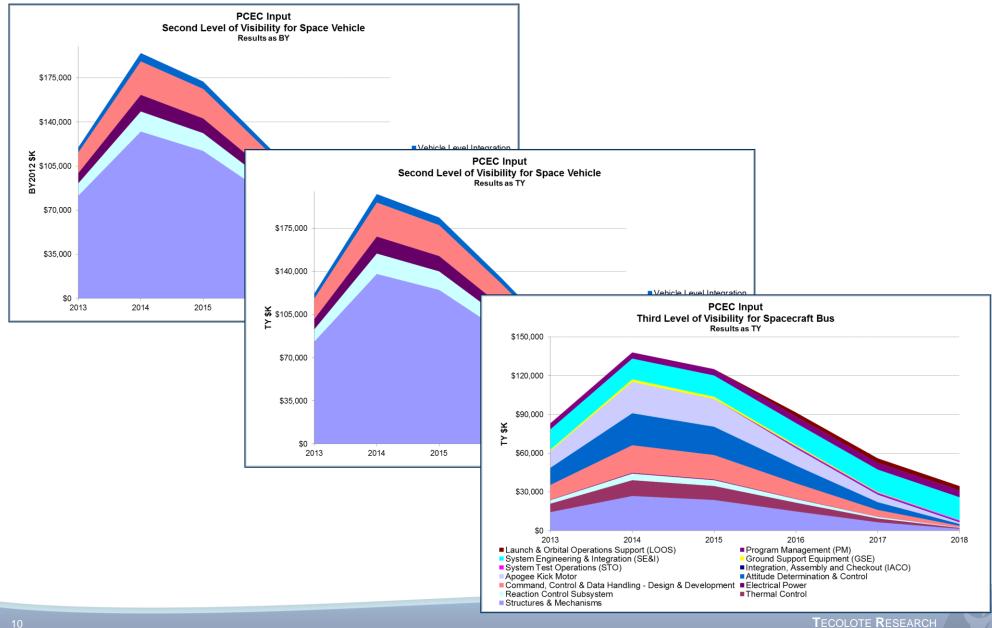
		<u>Structures</u>	& Mechanisms	
		Мај	or Inputs	
		Common Mu	lti-Variable Inputs	
Manufacturin	g Methods (2) Sig	, Mfg Techniques (25%)		
Engineering Ma	anagement (4) Sig	nificant Req. Changes (75	5%)	
N			The same of Company	
Funding			<u>Thermal Control</u>	
Tes			Major Inputs	
Integration (			Common Multi-Variable Inputs	
Pre-Developr		g Methods (2) Sig. Mfg Te		
	Engineering Ma	anagement (3) Mod Desig	gn Changes (50%)	
	N		Position Control Subsystem	
Large Iner			Reaction Control Subsystem	
Significant D	Tes		Major Inputs	
	Integration (		Common Multi-Variable Inputs	
	Pre-Developr		ods (1) Max. Mfg Techniques (6%)	
Quantity Next H			ent (1) Min Design Changes (6%)	
			(1) Existing "flight proven" design requiring no mods (5%)	
	Louve		lity (1) Funding is Assured - No Delays (25%)	
	Speci		ach (1) Minimum Testing, Qualification Using Simulation and Analysis (25	•
			xity (1) Minimal Major Interfaces Involving Multiple Contractors/Centers	· · ·
		Pre-Development St	udy (1) 2 or More Study Contractors in Phase A&B - Greater than 9 Months	of Study (25%)
	Quantity Next H			
·			PCEC Reaction Control Subsystem Specific Inputs	
		Thrust (	lbf) 1	
			Other Inputs	
		Quantity Next Higher Le	evel 1	
	•			

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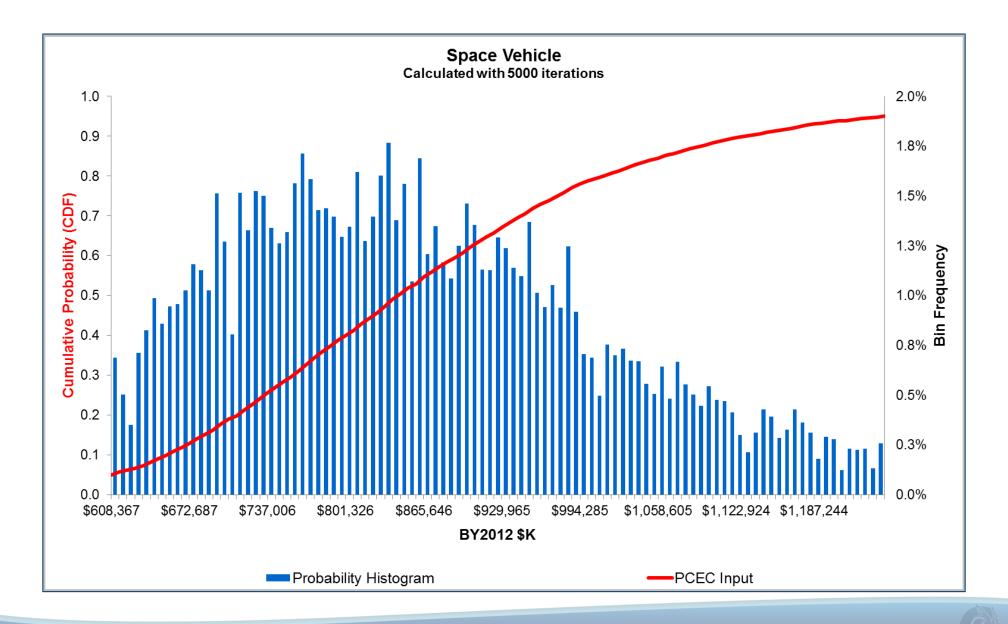
## **Cost Estimate Results**

	FY12\$K										]		
WBS Element		DDT&E		O&D	9	TH	Flt	Unit	Te	otal			
Spacecraft Bus	pacecraft Bus \$ 573,310.75 \$		159,774.11	\$	110,689.66	\$	153,366.80	\$	726,677.55				
Spacecraft Bus Subsystems	craft Bus Subsystems \$ 270,463.77 \$		159,774.11	\$	110,689.66	6 \$ 85,145.89 \$ 3			355,609.66				
Structures & Mechanisms		\$ 64,854.66	\$	41,389.95	\$	23,464.71	\$	\$ 18,049.78		\$ 82,904.44			
Thermal Control		\$ 27,625.82	\$	14,147.94	\$	13,477.88	\$	10,367.60	\$ 37,993.42				
Reaction Control Subsystem		\$ 12,045.22	\$	7,352.70	\$	4,692.52	\$	3,609.63	\$ 15,654.85		]		
Electrical Power and Distribu	ıtion	\$ 1,062.71	\$	149.97	\$	912.73	\$	702.10	\$	1,764.81			
Command, Control & Data Ha	andling	\$ 54,223.63	\$	38,982.58	\$	15,241.05	\$	11,723.88	\$	65,947.51			
Attitude Determination & Co	ontrol	\$ 59,252.82	\$	36,809.84	\$	22,442.98	\$	17,263.83	\$	76,516.65			
Apogee Kick Motor		\$ 51,398.91	\$	20,941.13	\$	30,457.78	\$	23,429.06	\$	74,827.98	]		
Spacecraft Bus System Integrati	ion	\$ 119,543.54	\$	-	\$	-	\$	19,185.27	\$	138,728.81			
Integration, Assembly and Cl	heckout (IACO)	\$ -	\$	-	\$	-	\$	1,332.19	\$	1,332.19	]		
System Test Operations (STC	D)	\$ 2,476.73	\$	-	\$	-			\$	2,476.73	]		
Ground Support Equipment (	(GSE)	\$ 6,121.88	\$	-	\$	-			\$	6,121.88	]		
System Engineering & Integr	ation (SE&I)	\$ 80,344.89	\$	-	\$	-	\$	12,014.72	\$	92,359.61	]		
Program Management (PM)		\$ 21,658.37	\$	-	\$	-	\$	5,838.36	\$	27,496.74			
Launch & Orbital Operations	Support (LOOS)	\$ 8,941.67	\$	-	\$	-			\$	8,941.67	]		
Fee			FY14\$K	·l									
Program Support	WBS Element	•				D&	.n	STH		El+ I	lnit		Total
Contingency										Flt Unit			
Vehicle Level Integration	Spacecraft Bus			\$ \$	596,784.95 281,537.91	\$ \$	166,316.06 166,316.06		115,221.84			<u> </u>	756,431.36 370,170.10
Total		Bus Subsystems		\$	67,510.13	\$	43,084.66			\$	<b>88,632.19</b> 18,788.83	<u> </u>	86,298.96
	Structures & Mechanisms Thermal Control		\$	28,756.96	\$	14,727.23		24,425.48 \$ 14,029.73 \$		10,792.10	<u> </u>	39,549.06	
		n Control Subsystem		\$	12,538.41	\$		\$	4,884.66 \$		3,757.43	+	16,295.84
		al Power and Distribution		\$	1,106.22	\$	156.12				730.85		1,837.07
		nd, Control & Data Handlir		\$	56,443.81	\$	40,578.72		-		12,203.92	+	68,647.73
		Determination & Control		\$	61,678.93	\$			15,865.09 23,361.90	Ś	17,970.70	_	79,649.62
					\$	31,704.88	\$	24,388.37		77,891.81			
	Apogee Kick Motor  Spacecraft Bus System Integration  Integration, Assembly and Checkout (IACO)			\$	124,438.25	Ś	21,730.30	\$	31,704.00	\$	19,970.81	· · ·	144,409.06
			\$	124,430.23	Ś		\$		\$	1,386.73	<del>-</del>	1,386.73	
		•	ut (IACO)	\$	2,578.14	Ś		Ś		Ś	1,300.73	Ś	2,578.14
	System Test Operations (STO)  Ground Support Equipment (GSE)		\$	6,372.54	\$		\$		\$		Ś	6,372.54	
			/c=Ω.I\	\$	83,634.61	\$		\$		\$	12,506.66	1	96,141.27
System Engineering & Integration (SE&I)  Program Management (PM)  Launch & Orbital Operations Support (LOOS)		\$	22,545.18	-		\$		\$	6,077.42	· ·	28,622.59		
		\$	9,307.78	\$		\$		\$	0,077.42	Ś	9,307.78		
	Fee Fee		011 (1003)	\$	48,717.14	*		\$		\$	13,032,36	· ·	61,749.50
	Program Su	innort		\$	40,597.62		- \$ - \$					· · · · ·	51,457.92
	Contingen			\$	81,195.23	\$		\$		\$	21,720.60	· -	102,915.83
		yel Integration			20,298.81			Ś		Ś	5,430.15	<u> </u>	25,728.96
	Total	.ctegration		Ś	596,784.95	-		T	115,221.84	*	159,646.40	+	756,431.36
				- C	506 78/1 05	Ś	166,316.06	<b>C</b>	115 771 8/		150 6/16 //0	Ś	

## **Time-Phased Results**



## **Risk Results**



## In Conclusion...

PCEC/ACEIT combines the power of several frameworks into a single tool for analysis and reporting

PCEC in ACE Space Vehicle Subsystem model is currently in Beta, but will soon be available to all NASA via CAD

PCEC in ACE results (deterministic and probabilistic) track to PCEC published results

All NASA Organizations have ACEIT licensed

Additional capabilities are being considered for future versions of the PCEC/ACEIT tool - send in feedback

# **Thank You**

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